

Compensation Filter for Aircraft Control System

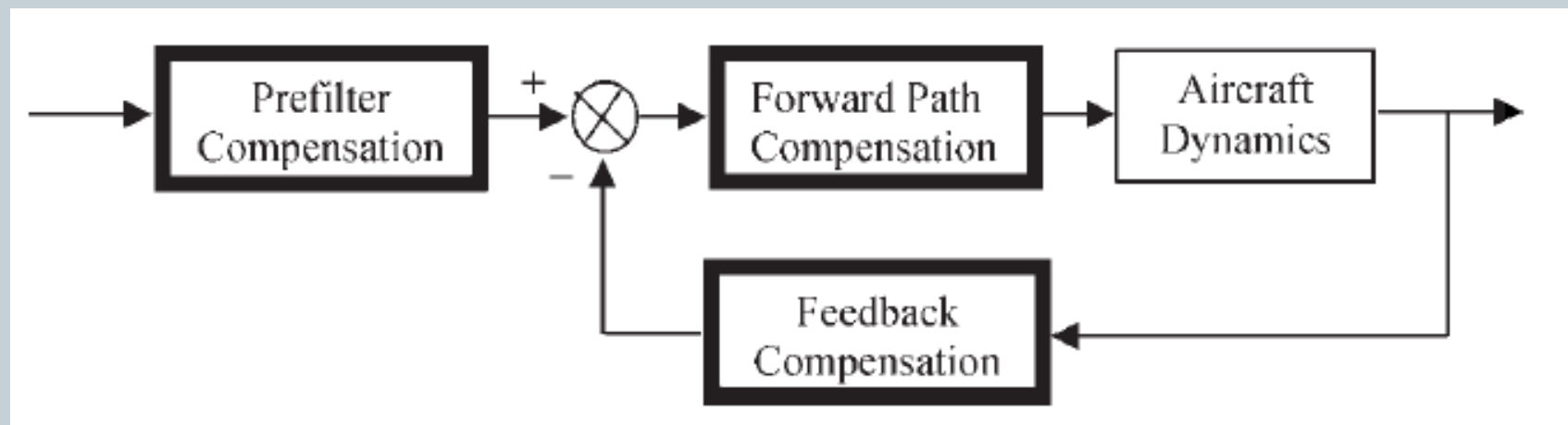


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- A powerful tool available to the control system designer is a **compensation filter**. Compensation filters can take a variety of forms and are **very effective in tailoring the aircraft response**
- In a generic feedback control system, a compensation filter can be located in generally three possible positions
 - i) **Pre filter compensation**
 - ii) **Forward path compensation**
 - iii) **Feedback compensation**
- **Pre-filter compensation** modifies the closed loop transfer function **directly**, while **forward path and feedback compensation** modify the **forward path and feedback path transfer functions**, which are inputs for the root locus.



Possible locations of compensation filters



- Pre-filter compensators generally use the principle of **cancellation of an undesirable closed-loop pole with a Pre-filter zero, and cancellation of a closed-loop zero with a Pre-filter pole.**
- Forward path and feedback path compensation filter allows **modification of the root locus through the addition of poles and zeros in desire location.**
- The filters are generally implemented in the flight control computer and can be thought of as a added software for digital systems and additional circuitry for analog systems.

Lead compensators



- Magnitude of zero is high than the magnitude of pole.
- Lead compensators are generally used to quicken the system response by increasing natural frequency and/or decreasing time constant.
- Lead compensator shift the root locus towards the left hand side of the complex s plane.
- It help to increase the stability of the system as we all speed of response (shifts gain crossover freq to higher value).
- Additional positive phase increases the phase margin and thus increases the stability of the system.
- It minimize system overshoots and also increases the velocity constant of a system (K_v).

note: washout circuit is an example for lead network



- Lead compensator has general form of

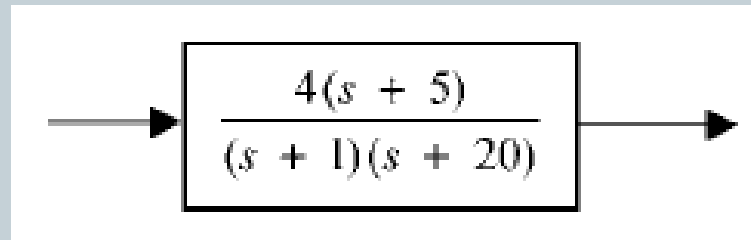
$$TF_{lead\ compensator} = \frac{b(s + a)}{a(s + b)} \quad a < b$$

- The practical limit in choosing the pole and zero for the lead compensator is $b < 10a$.
- A common application of lead compensators is to cancel a pole at $s = -a$, which is slowing the time response or causing the system to be unstable.

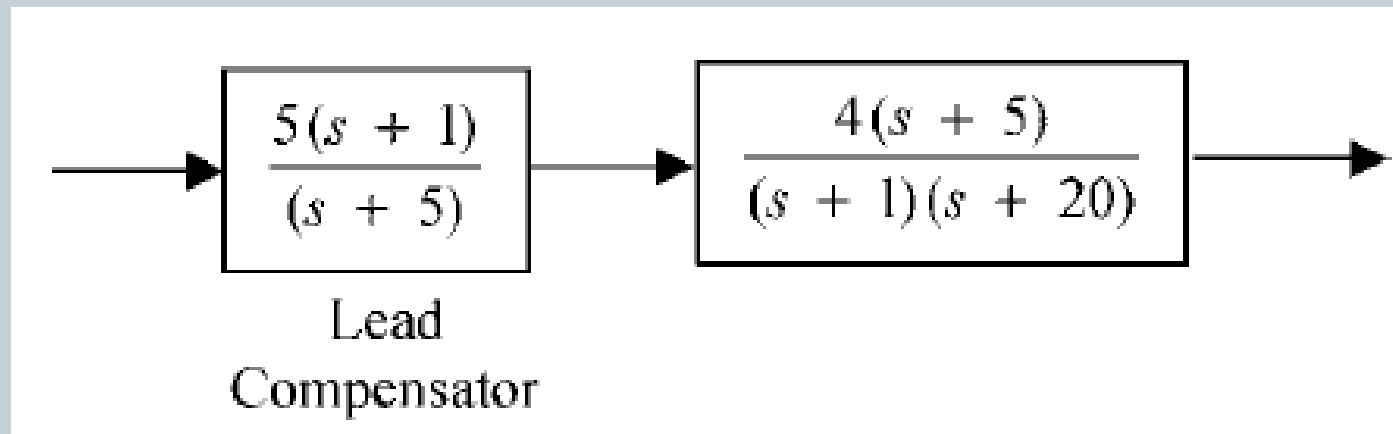
Example



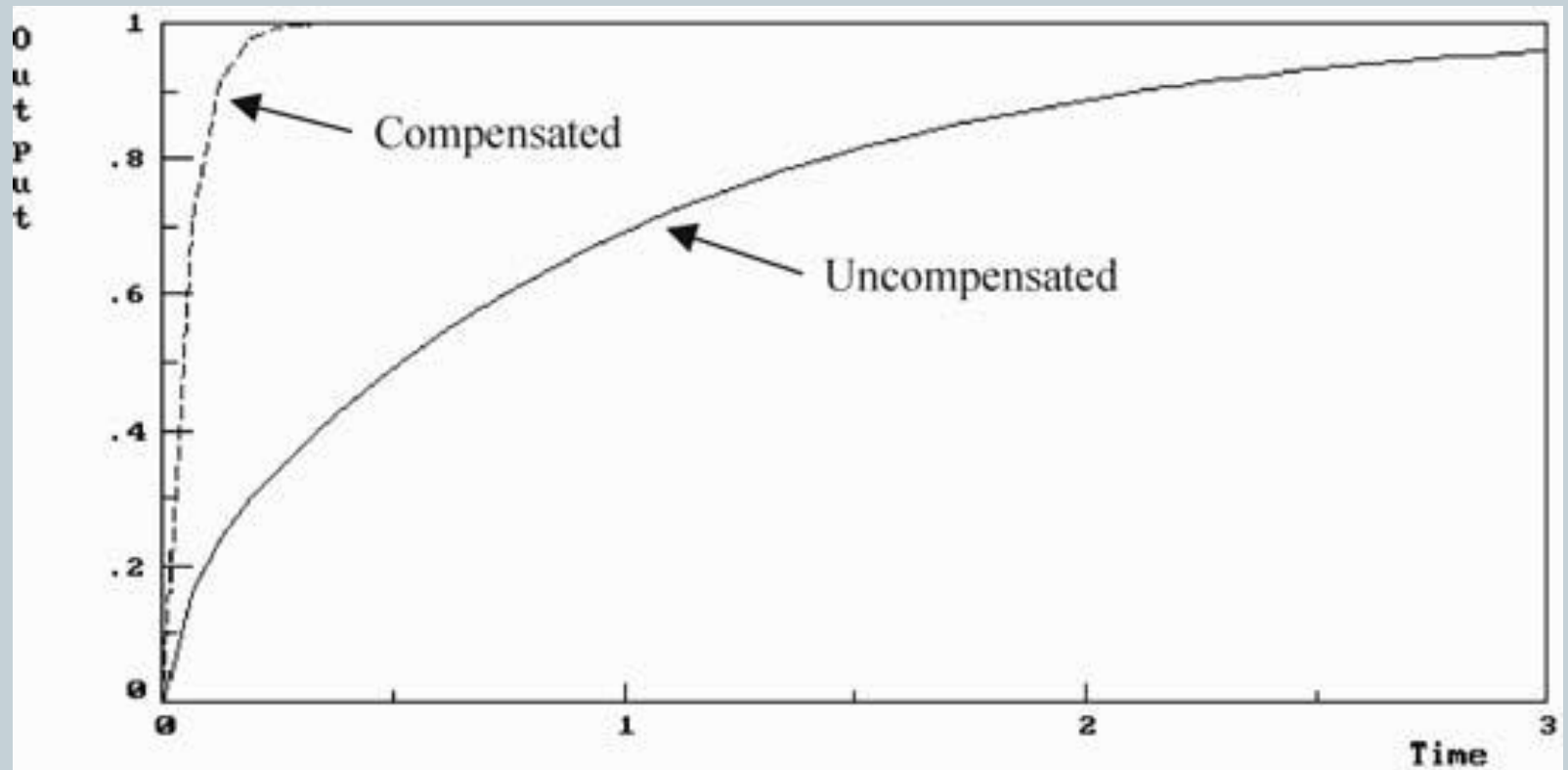
- Design a pre filter lead compensator to decrease the time constant of the following system to less than 0.2 s.



- The time response of the system will be composed of two components, each directly dependent on the characteristics of the two poles. Notice that the pole at $s=-1$ has a time constant of approximately 1 s. The pole at $s=-20$ has a time constant of 1/20th of a second and is not a problem. A simple lead compensator can be used to cancel the problem pole.



- Notice also that we have cancelled the zero at $s = -5$ with the pole on the lead compensator.

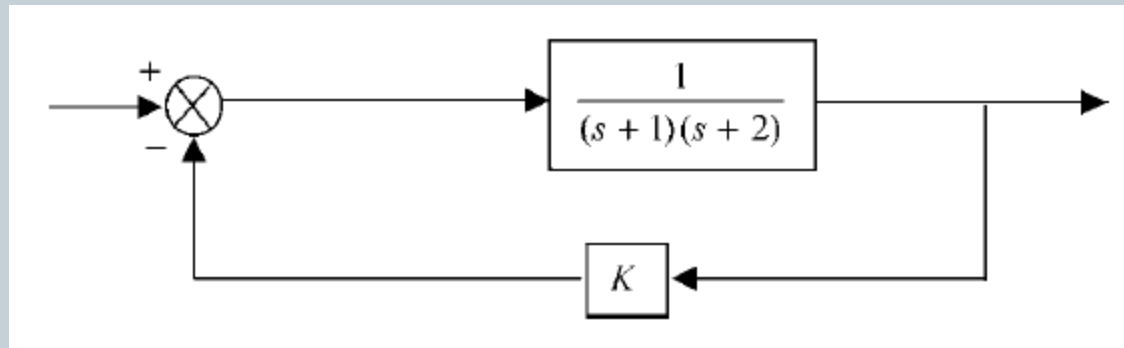


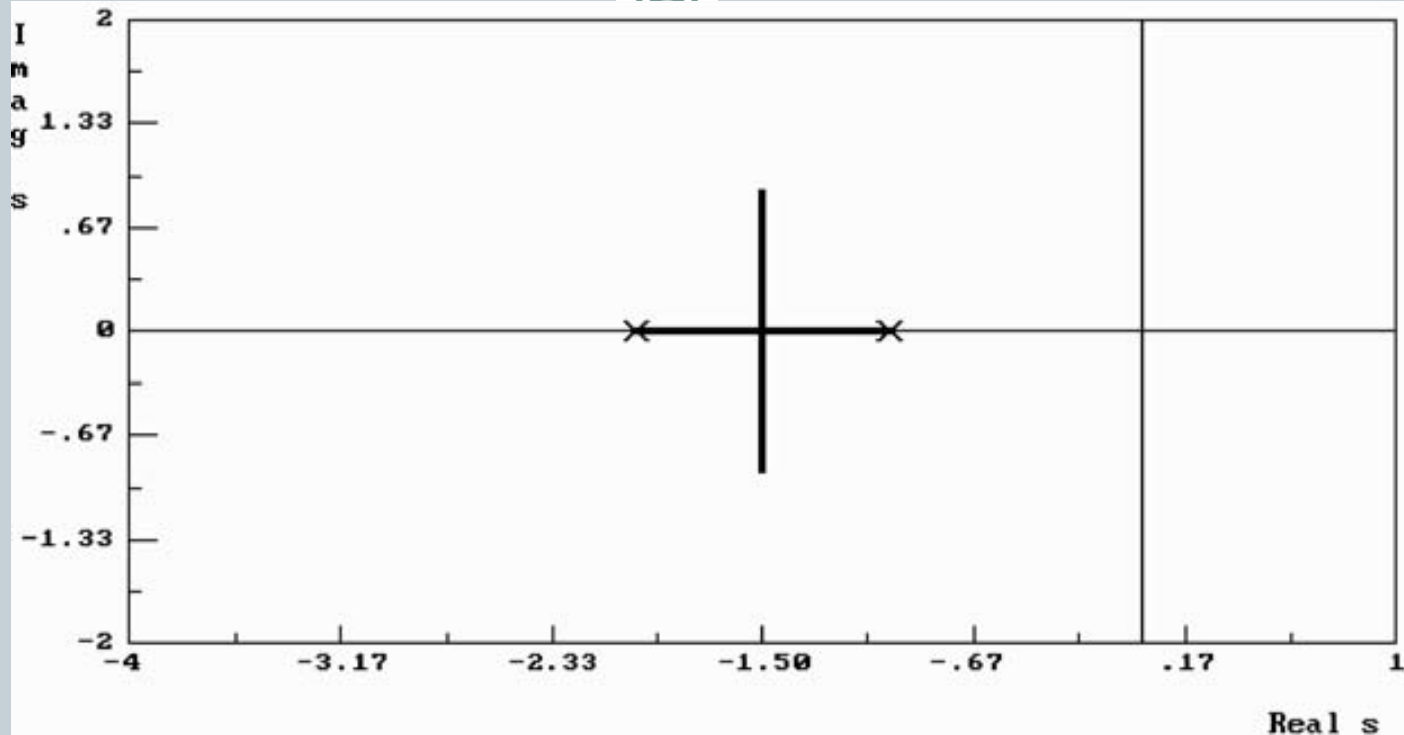
Time response characteristics with and without the lead compensator

Example 2

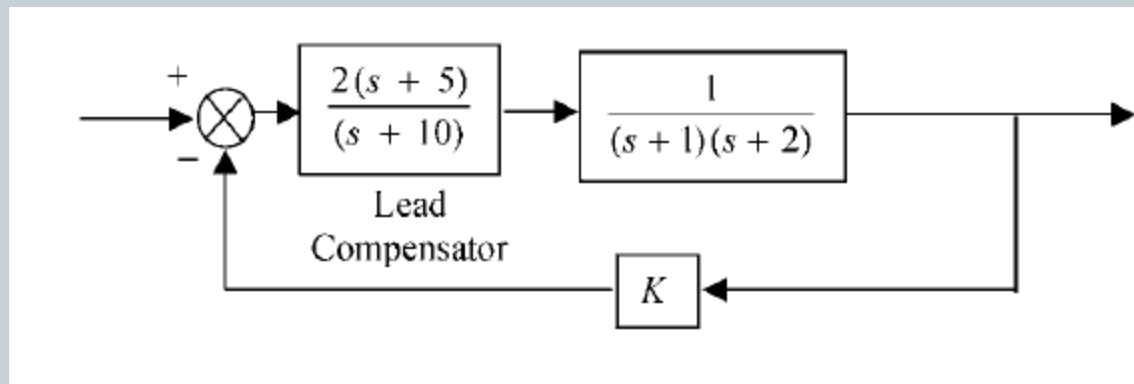


- Design a forward-path lead compensator for the following system that will shift the root locus to the left.



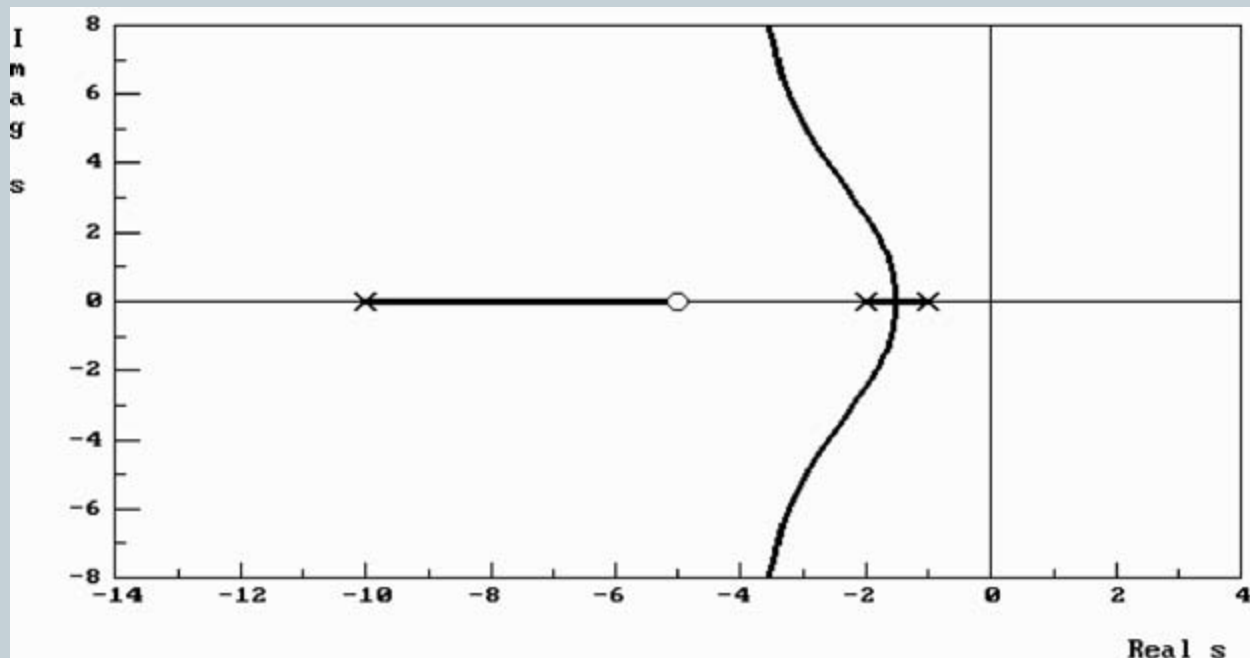


- We add a forward-path compensator that will place a pole and zero to the left of the two open-loop poles. The root locus is attracted to the compensator zero at $s = -5$.





- The compensated root locus is presented next. Notice that the vertical branches have shifted to the left.



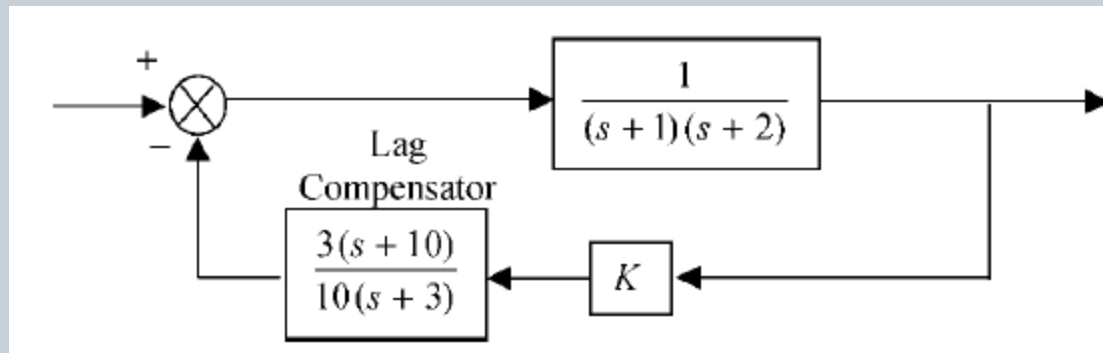
Lag compensator



- Lag compensator shift the root locus towards the right hand side of the complex s plane.
- It help to decrease the stability of the system as well as to reduce speed of response(aircraft response).
- It also helps to increases steady state accuracy. Increasing steady state accuracy helps to induce instability.
- With lag compensation, a pole is added to the right of a zero. The pole may be used to cancel a zero, or it may be used to shift the root locus to the right.
- Lag compensation may also reduce the steady-state error of a system.

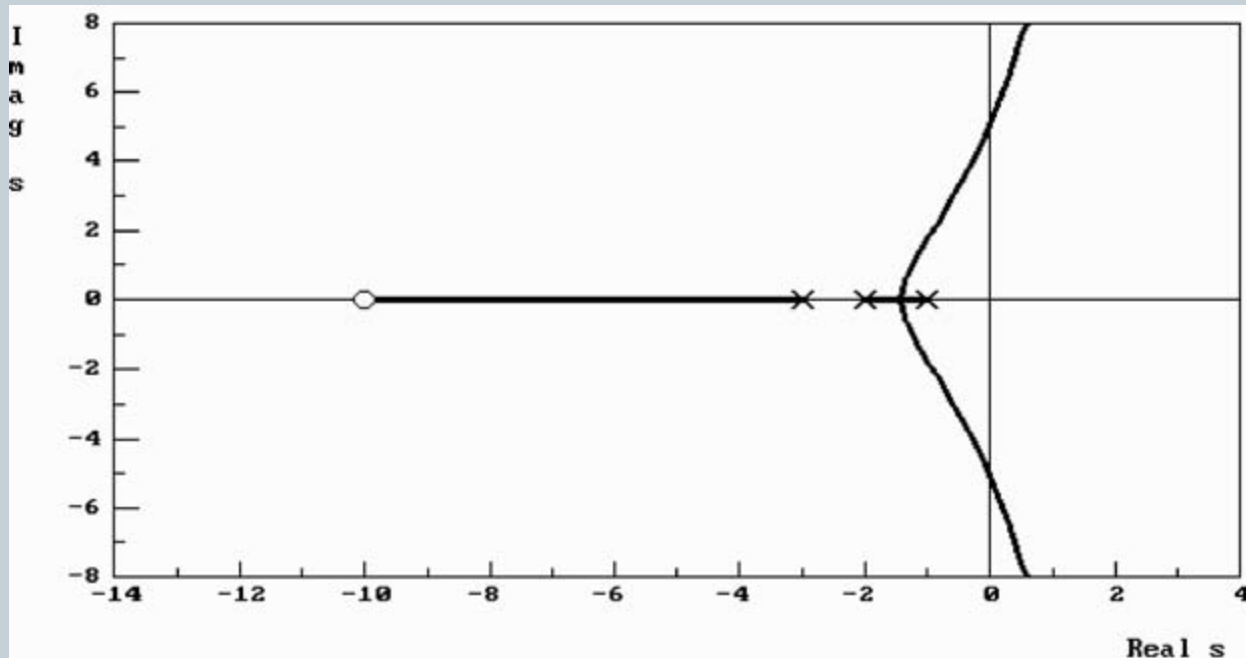
$$TF_{\text{lag compensator}} = \frac{b(s+a)}{a(s+b)} \quad a > b$$

- The lag compensator pole is placed at $s=-3$ to repel the root locus to the right.





- The root locus for the compensated system is presented next. Notice that the root locus branches have shifted to the right.



Lead-Lag Compensator



- A lead-lag compensator normally adds **two zeros that are fairly close together and that provide a powerful attraction for root locus branches.**
- In many cases, the **useful gain range** (before a system goes unstable) can be **increased using a lead-lag compensator.**
- Another common use of lead-lag compensators is the **attenuation of a specific frequency range** (sometimes called a notch filter).
- For example, an aircraft structural resonant frequency can be filtered out with a lead-lag compensator if a feedback sensor is erroneously affected by that frequency



- A lead-lag compensator has the general form

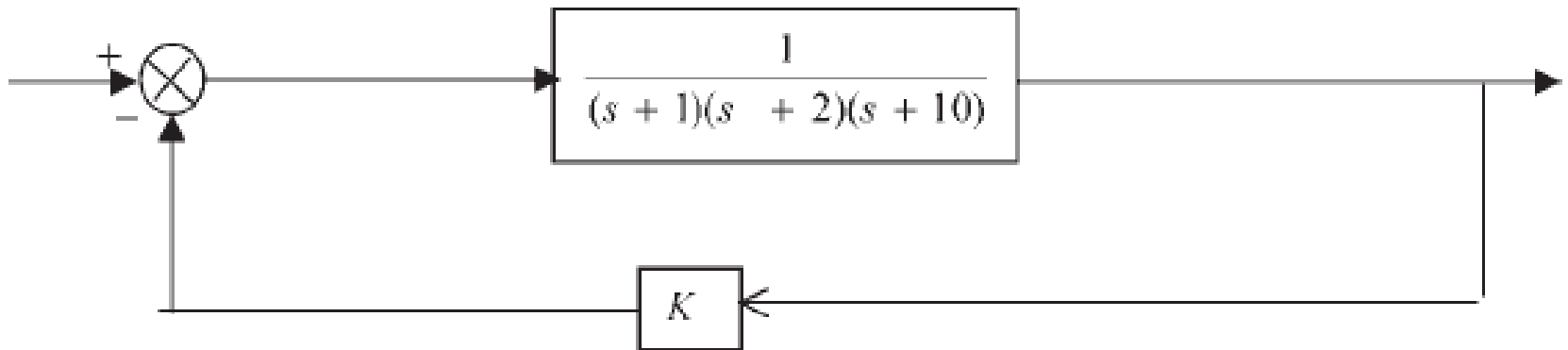
$$TF_{lead\ lag\ compensator} = \frac{bd(s+a)(s+c)}{AC(s+b)(s+d)} \quad a > b, a < c, c < d$$

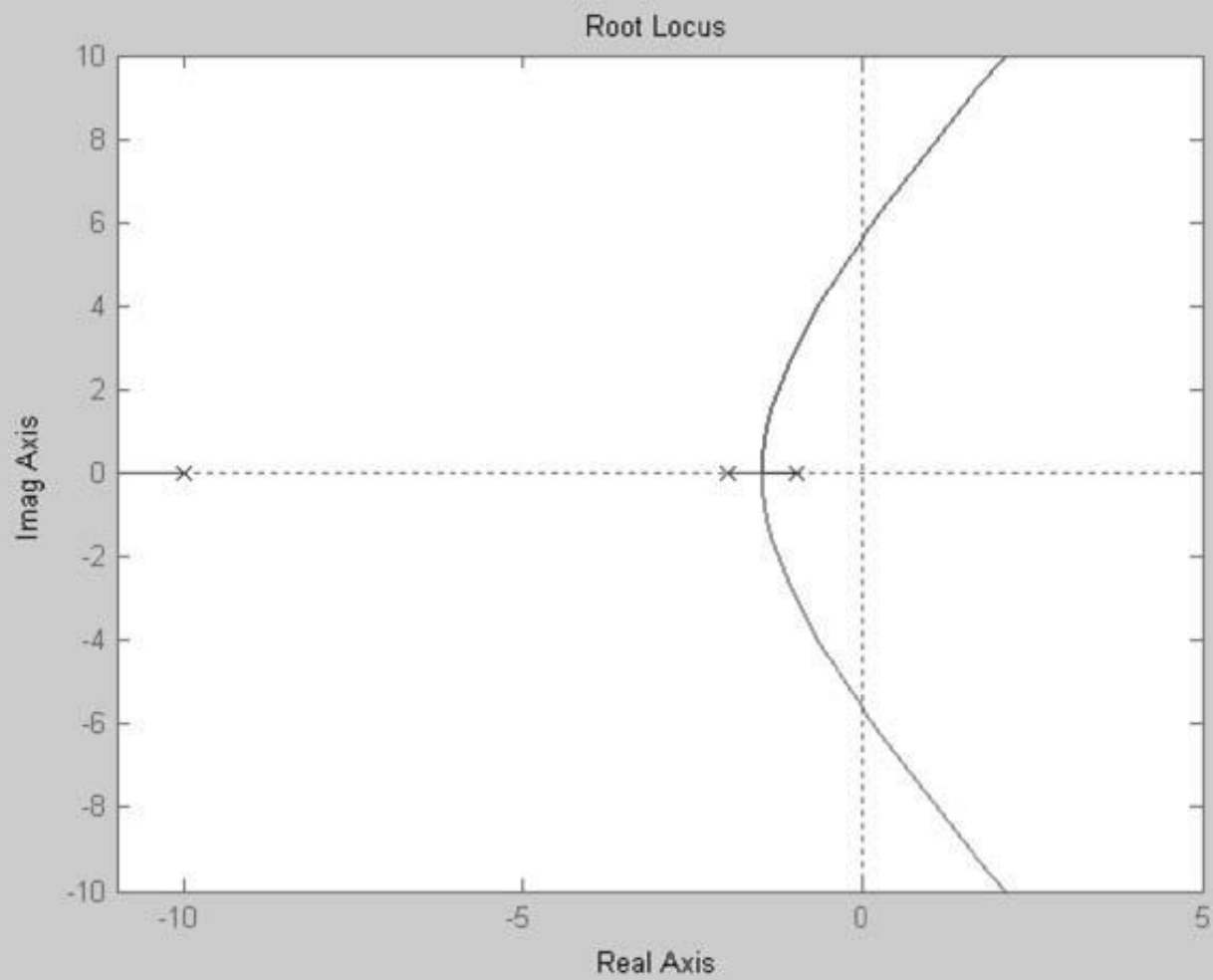
$$\frac{(s+a)}{(s+b)} \text{ component represent lag filter}$$

$$\frac{(s+c)}{(s+d)} \text{ component represent lead filter}$$



- Starting with the following system, design a lead-lag feedback path compensator that will provide for stable roots at higher values of K .

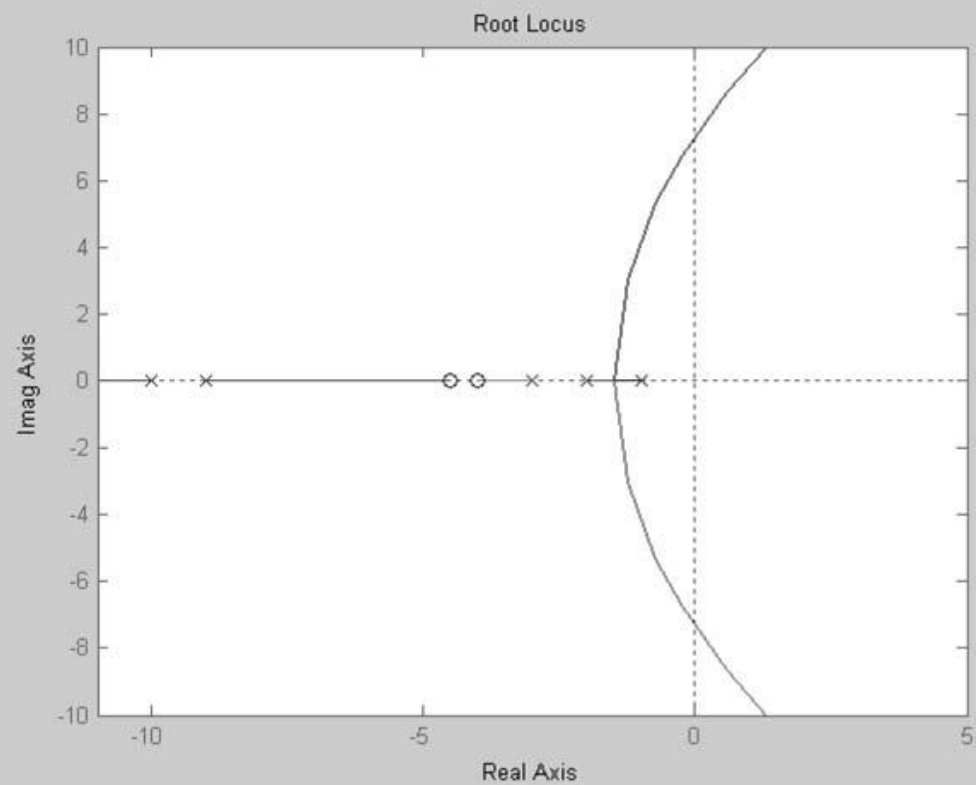
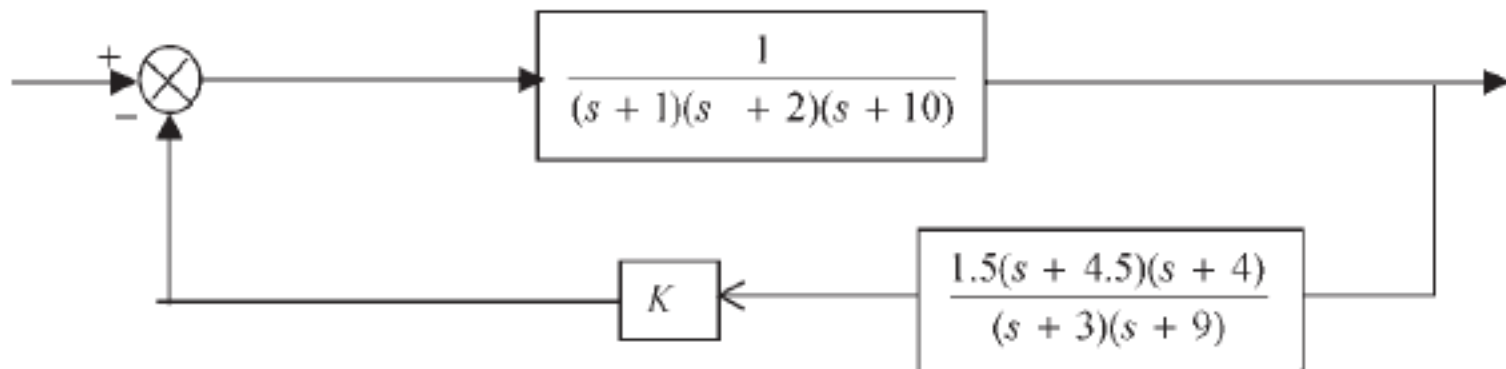






- The two complex branches of the root locus go unstable for values of K greater than 391. To allow for a larger range of stable gain values, we add the feedback path lead-lag compensator that places two zeros to the left of the pole at 2.

$$\frac{1.5(s + 4.5)(s + 4)}{(s + 3)(s + 9)}$$





- The two complex branches of the compensated root locus now go unstable for K values greater than 600. Thus, we have gained a larger range of stable values of K .



Thank you